Load Sources: Aerodynamic / neitia loads Manoevers and gusts Emergency landing inertias Cabin pressurisation loads 5 fatigie loading landing loads (> high loading + fatigue loading Engine loado 4 thust, weight, torque and gyroscopic effects Thermally induced loads (> differential thermal expansion Impact / crosh loads Types of Load: Tension - tie bars and cables effective corriers (> one strand fracture + equal failure , main issues for tension loaded structures are fracture & fatigue Compression - thin walled tube on 2nd MoA maximised for given amount of material critical due to buckling depends on I_{∞} and Efatigue lass of concern as cracks don't propagate under compression Shear - conside by tensile & compressive members at 45° Panels also good at carmying shear by resisting shape change Bending - I beam effective to resist bending due to 1/2

Bending with shear

tusting moment carried effectively by shear forces around closed structure - a box/tube with continuous skin is best Torsion alternatively crossed brace members

Types of Airframe Construction:

- Frame with non-structural covering
- → all loads carried by frame structure → skin assumed not to carry any load except air pressure
- used for early aircraft with You loads

- Monocoque

- → single skin with no supporting frame → all bods carried by skin
- -> poor for large size due to buckling

- Stressed Skir

Stunger - styfening element } libs - divide skin into small panels to resist buckling

Frame

-> load divided between skin & stringers

Shear Resistant Structures

→ Panels carry ohear without buckling → Stiffners divide pannels up to avaid buckling

Tension Field Structures

compression carried by stiffners Orgonal tension carried by parels

Functions of Structural Members: duride up skin tensile smooth terrile loads aers Surface Stungers - compressive Skin compressive loads bending stiffness add stiffness shear loads sealed to skin to resist endosure buckling - Frames & Ribs aganist buckhing (fuselage) (ming) Maintain cross section shape panels to reduce distribute discrete Shear buckling loads into rest of Structure